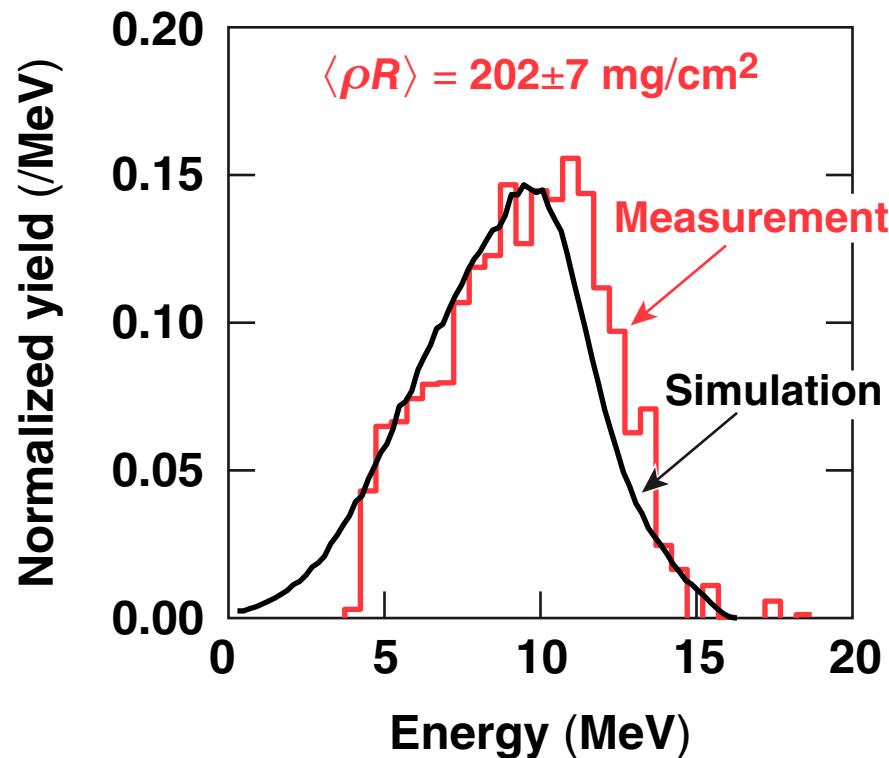


# Performance of Direct-Drive Cryogenic Targets on OMEGA



V. N. Goncharov  
University of Rochester  
Laboratory for Laser Energetics

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American Physical Society  
Division of Plasma Physics  
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## Summary

**High-areal-density ( $\rho R > 200 \text{ mg/cm}^2$ ), low-adiabat cryogenic-fuel assembly has been achieved on OMEGA**



- ICF ignition designs rely on low-adiabat fuel assembly
- Maintaining the low level of fuel adiabat during the implosion requires an accurate account for all sources of shell heating, including
  - shock heating
  - preheat due to radiation and suprathermal electrons
  - short-wavelength perturbation growth
- Effects of nonlocal thermal transport are important to model the shock heating
- Areal-density close to 1-D prediction is achieved when the shocks are accurately timed and the suprathermal-electron preheat source is mitigated

# Collaborators

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**T. C. Sangster, P. B. Radha, R. Betti, T. R. Boehly, T. J. B. Collins,  
R. S. Craxton, J. A. Delettrez, R. Epstein, V. Yu. Glebov, S. X. Hu,  
I. V. Igumenshchev, R. Janezic, J. P. Knauer, S. J. Loucks, J. R. Marciante,  
J. A. Marozas, F. J. Marshall, D. N. Maywar, R. L. McCrory, P. W. McKenty,  
D. D. Meyerhofer, S. P. Regan, R. Roides, W. Seka, S. Skupsky,  
V. A. Smalyuk, J. M. Soures, and C. Stoeckl**

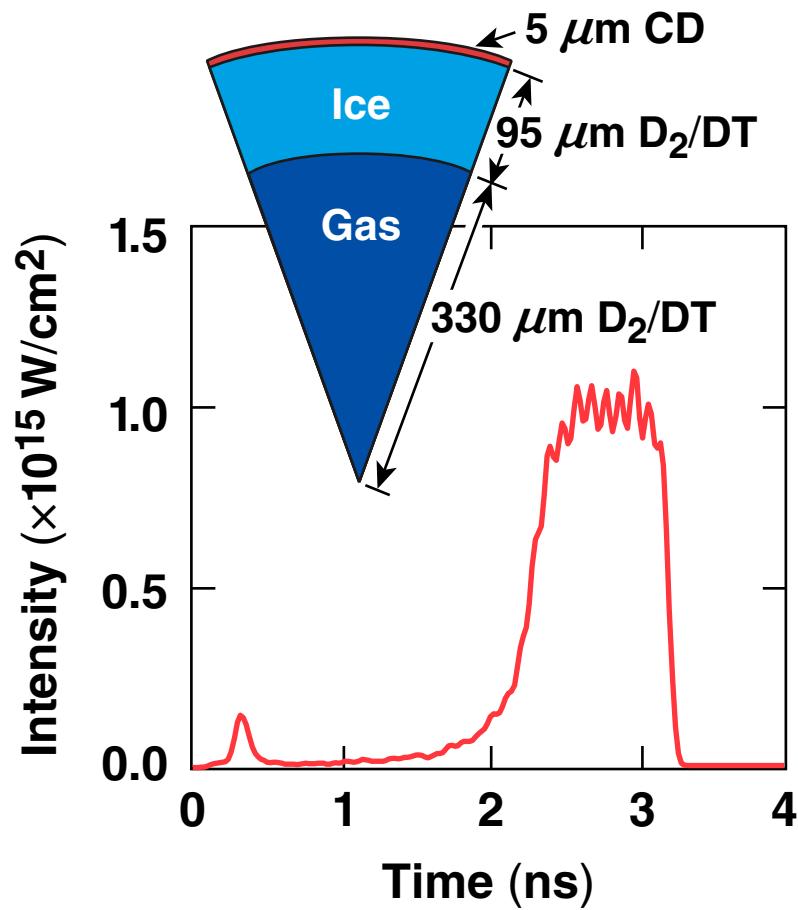
**University of Rochester  
Laboratory for Laser Energetics**

**D. Shvarts**  
**Nuclear Research Center  
Negev, Israel**

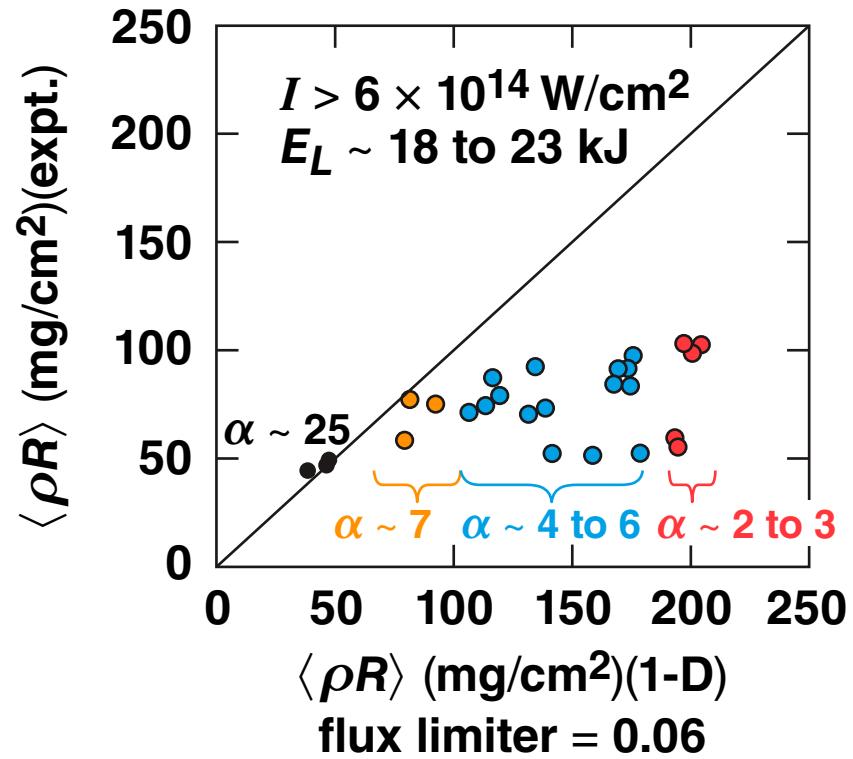
**J. A. Frenje, R. D. Petrasso, and C. K. Li**  
**Massachusetts Institute of Technology**

**W. Manheimer and D. Columbant**  
**Naval Research Laboratory**

To study physics of low-adiabat fuel compression,  
a series of cryogenic experiments has been  
performed on OMEGA



$$\alpha = \frac{P}{P_{\text{Fermi}}}$$



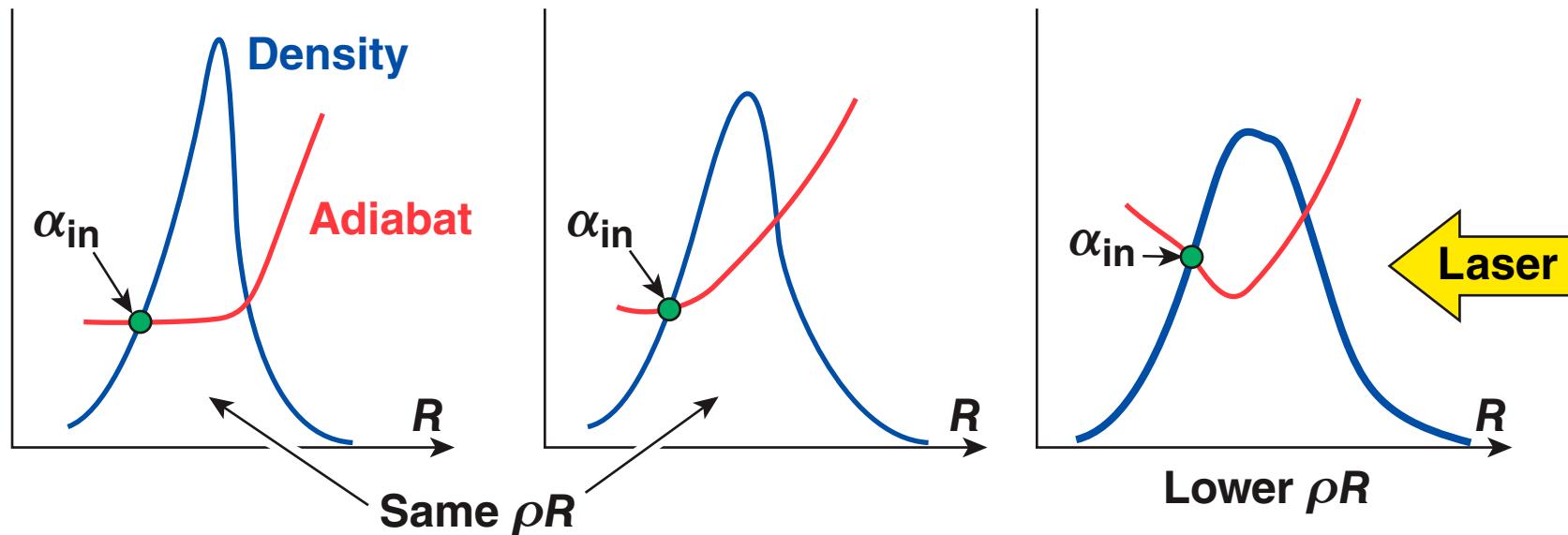
F.J. Marshall *et al.*, Phys. Plasmas **12**, 056302 (2005).  
T.C. Sangster *et al.*, Phys. Plasmas **14**, 058101 (2007).

# For a given laser energy, areal density is sensitive only to the shell adiabat\*



- $\rho R = \frac{2.6}{\alpha_{in}^{0.54}} E_{MJ}^{1/3}$

- Degradation in  $\rho R$  can be due only to the excessive adiabat increase (extra shell heating).

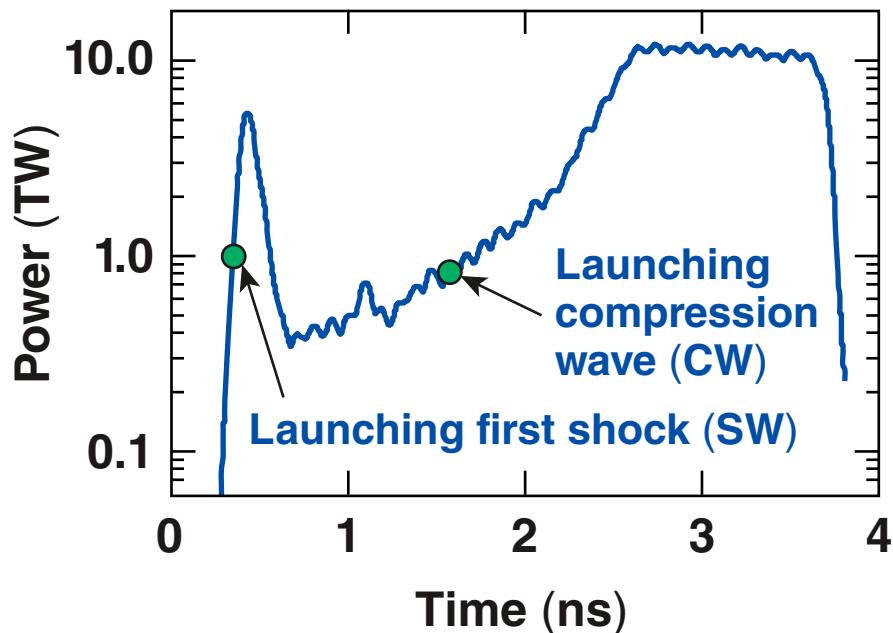


## Shock Timing

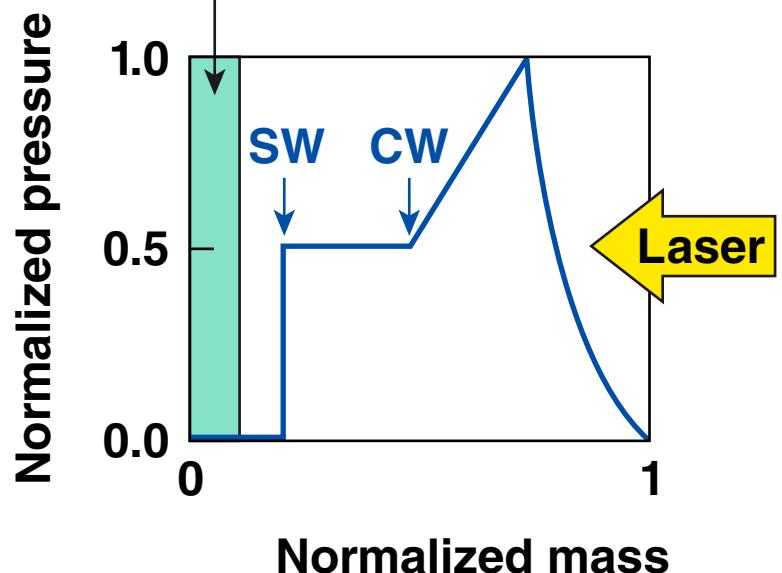
The main sources contributing to the adiabat degradation include hydrodynamics, radiation and electron preheat, and small-scale nonuniformity growth



- Timing of hydrodynamic waves



CW and SW/RW must coalesce inside 10% initial mass region

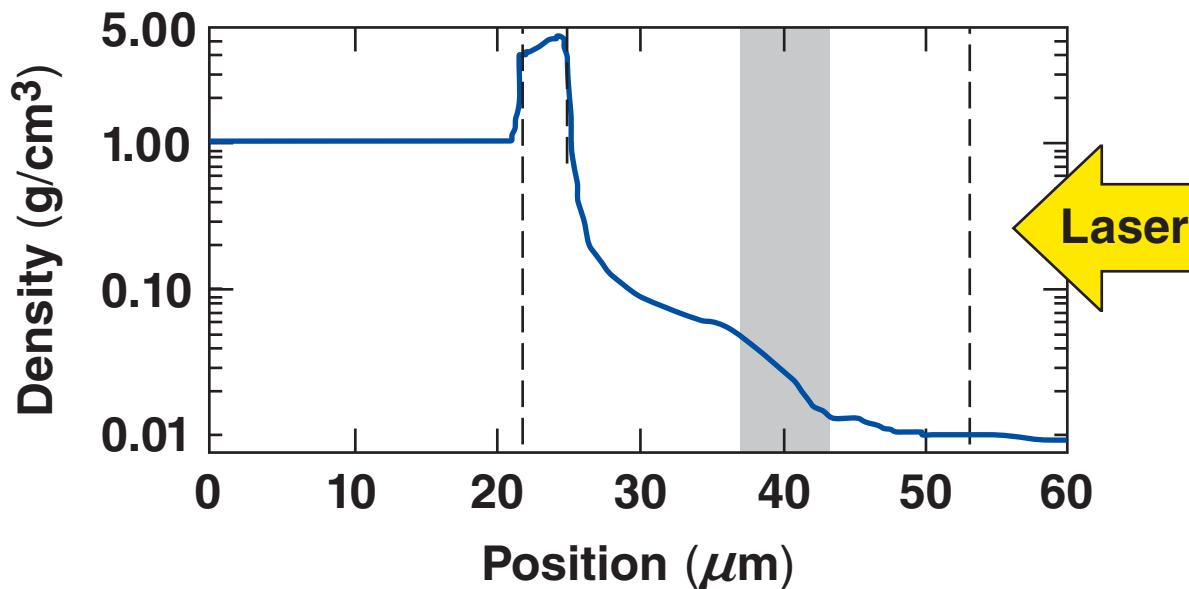


$$\frac{\Delta t_{\text{shock}}}{t_{\text{shock}}} < 5\%, \quad t_{\text{shock}} \sim E_p^{-1/2} \Rightarrow \frac{\Delta E_p}{E_p} < 10\%$$

# A flux-limited thermal transport model\* produces sharp density near critical surface



- $q_{SH} = -\kappa \nabla T$      $q_{FS} = n T V_T$
- $q = \min(q_{SH}, f q_{FS})$
- $0.04 < f < 0.1$



A sharp gradient leads to low laser absorption at the beginning of the pulse

## Shock Timing

# Laser coupling depends on thermal conduction modeling



- Flux-limited Spitzer conduction model with constant flux limiter is not sufficiently accurate
- New nonlocal model solves Boltzmann equation with Krook's collision operator\*

$$v \frac{\partial f}{\partial x} + \frac{eE}{m} \frac{\partial f_0}{\partial v_x} = -\nu(v)(f - f_0), \quad f = f_0 - \int^x \left( \frac{\partial f_0}{\partial x} + \frac{eE}{T} \frac{\partial f_0}{\partial v_x} \right) e^{-\int^{x'}_{x''} \frac{dx''}{\lambda_{ei} \cos \theta}} dx'$$

- To conserve the number of particles and thermal energy, the coefficients in  $f_0$  are renormalized

$$n' = n + \delta n, \quad T' = T + \delta T$$

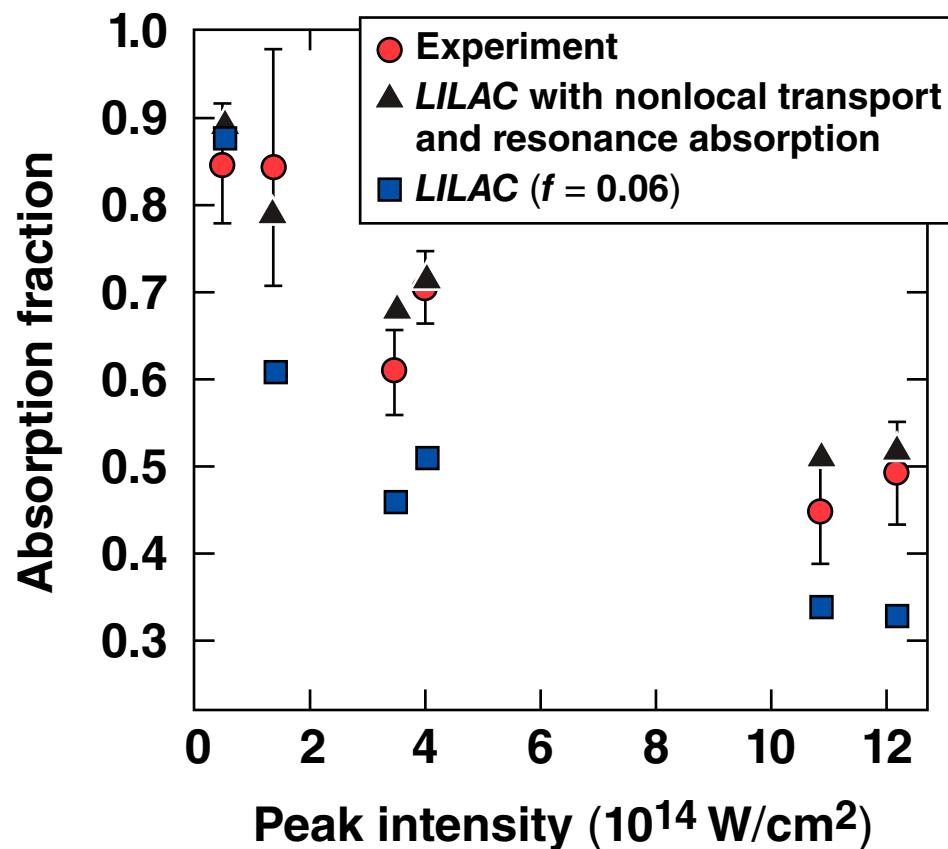
$$\text{in the limit } \frac{\lambda_{ei}}{L_T} \ll 1, \quad \delta n \sim O\left[\left(\frac{\lambda_{ei}}{L_T}\right)^2\right]$$

## Shock Timing

Simulations using the resonance absorption and new nonlocal transport model agree well with experimental data\*



- 200 ps Gaussian pulse

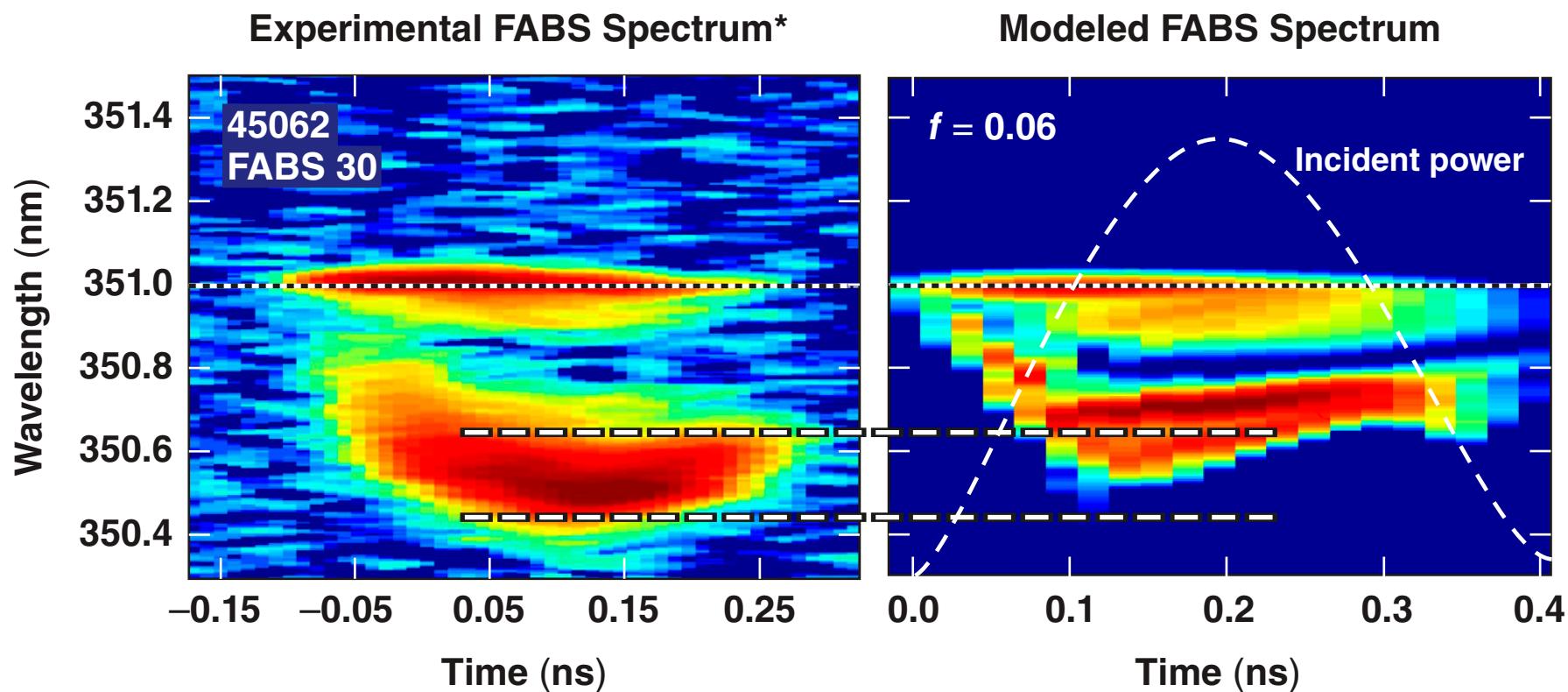


## Shock Timing

The nonlocal model is required to explain observed spectral shifts in scattered light



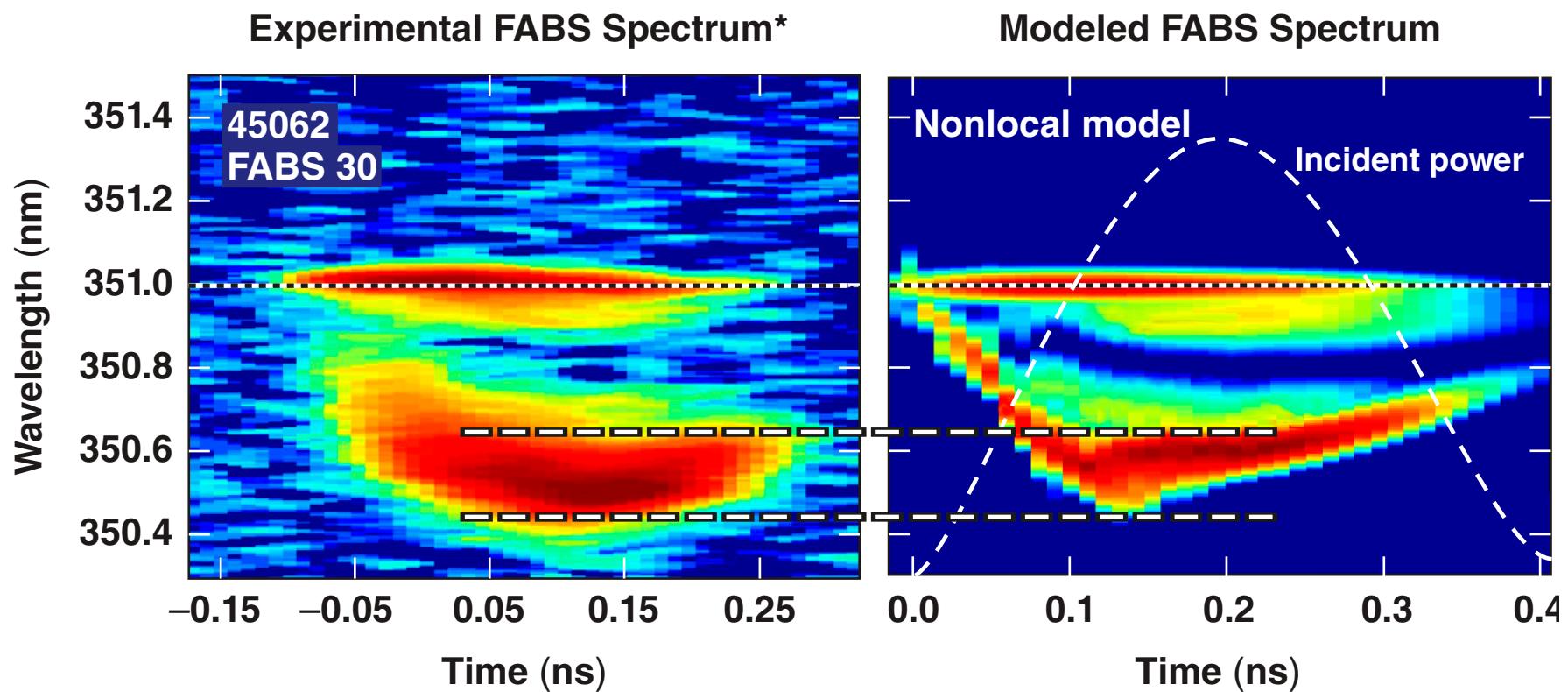
$$\text{Spectral shift}^1 \Delta\omega = \frac{\omega_0}{2c} \int \left(1 - \frac{n_e}{n_c}\right)^{-1/2} \frac{\partial}{\partial t} \left(\frac{n_e}{n_c}\right) ds$$



<sup>1</sup>T. Dewandre, J. R. Albritton, and E. A. Williams, Phys. Fluids 24, 528 (1981).  
<sup>2</sup>D. Edgell, NO6.00009, this conference

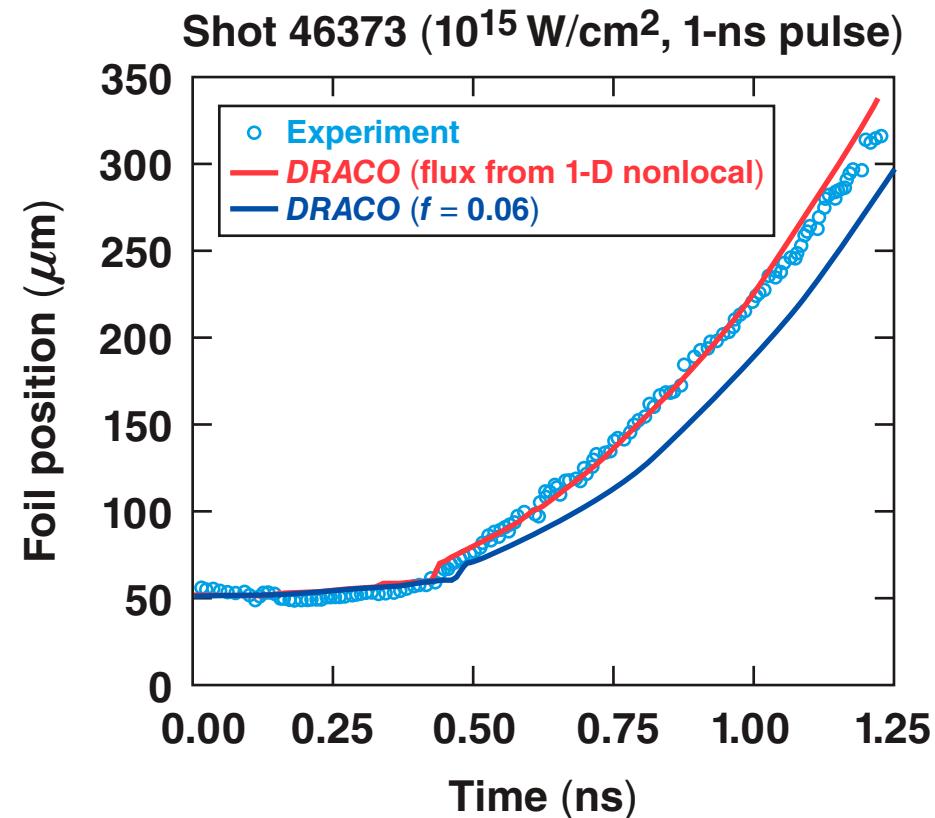
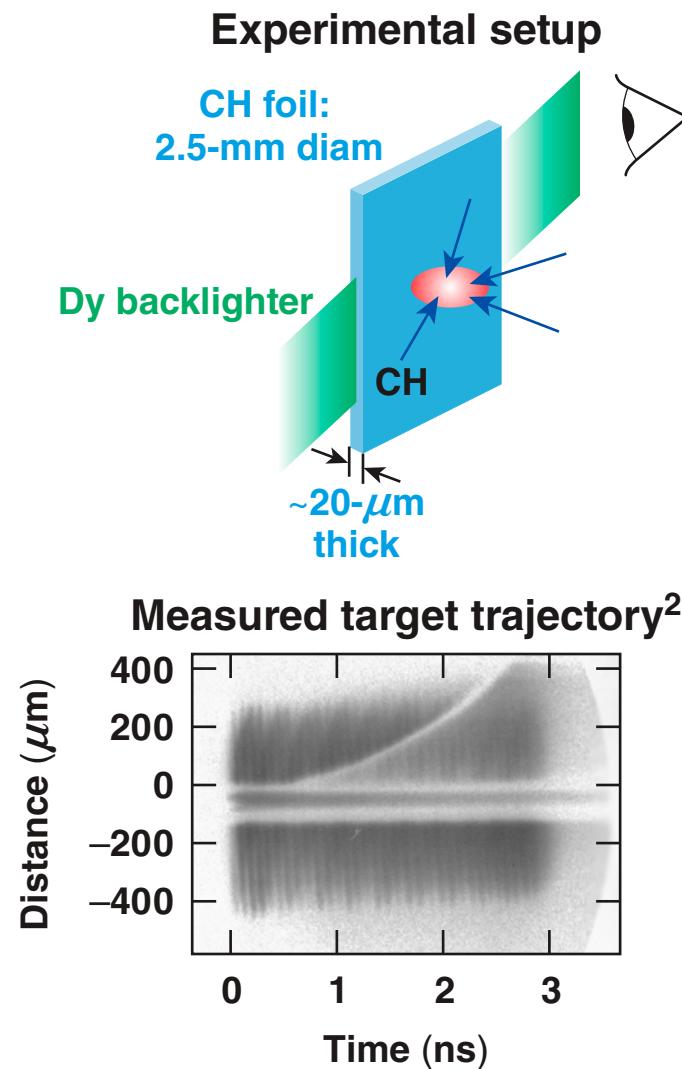
## Shock Timing

The nonlocal model is required to explain observed spectral shifts in scattered light



## Shock Timing

Measured foil trajectory is in agreement with the results of simulations<sup>1</sup> using the nonlocal transport model

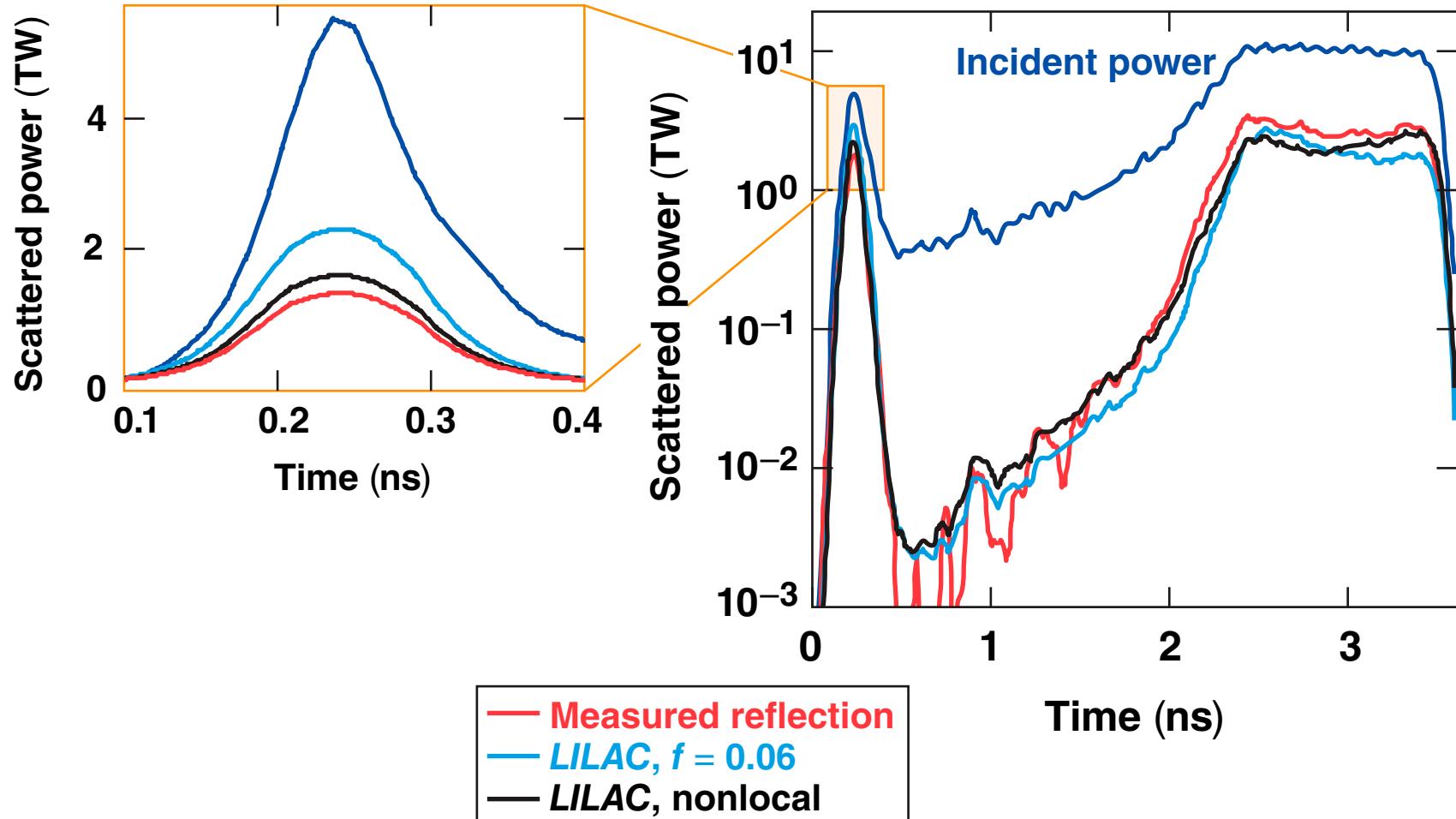


<sup>1</sup>S. Hu (PO6.00014)

<sup>2</sup>V. Smalyuk et al., presented at the 37th Anomalous Absorption Conference, Maui, HI, 27–31 August 2007

## Shock Timing

Time-resolved laser-absorption measurements\* are in good agreement with the results of the nonlocal model

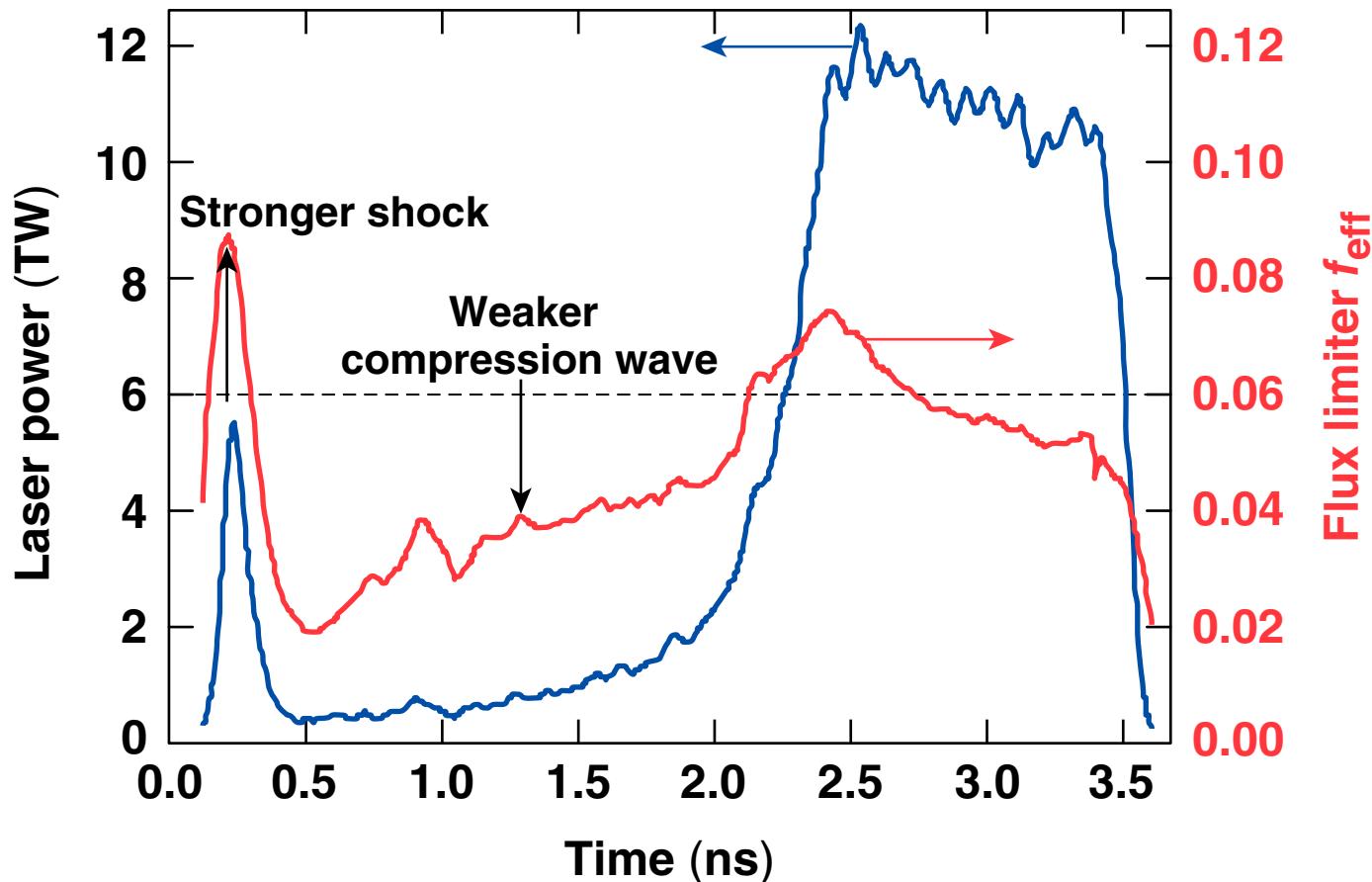


## Shock Timing

The effective flux limiter is calculated using the results of the nonlocal model

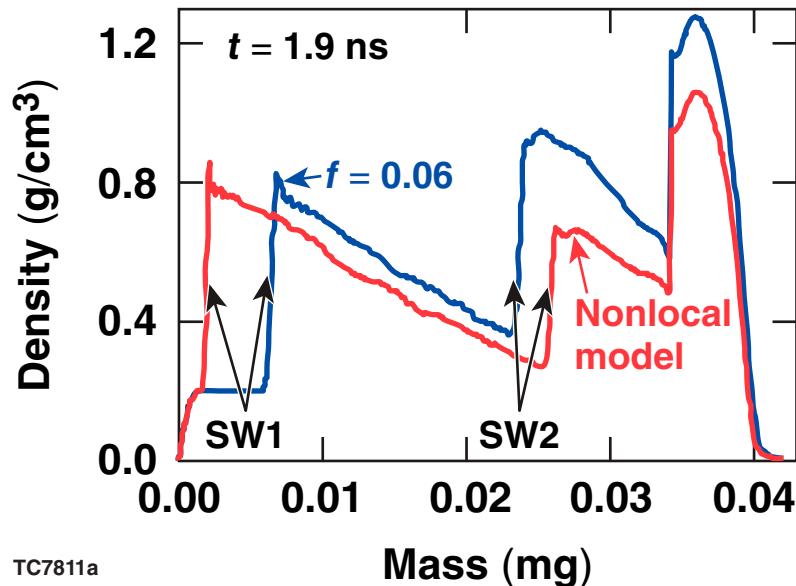
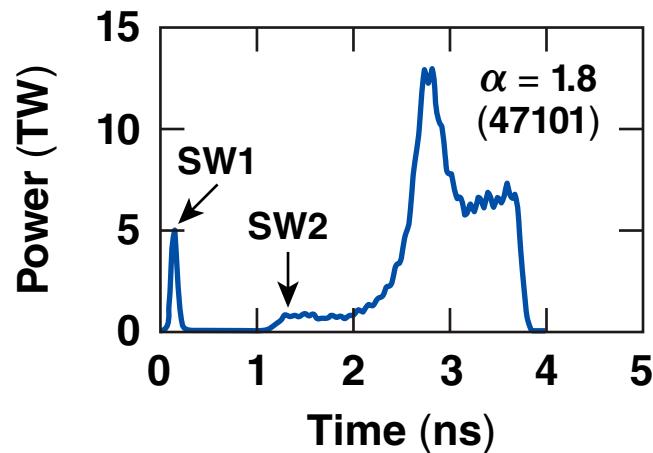


$$f_{\text{eff}} = \max\left(\frac{q_{\text{NL}}}{q_{\text{FS}}}\right) \quad q_{\text{NL}} = \text{nonlocal flux} \\ q_{\text{FS}} = \text{free-stream flux}$$

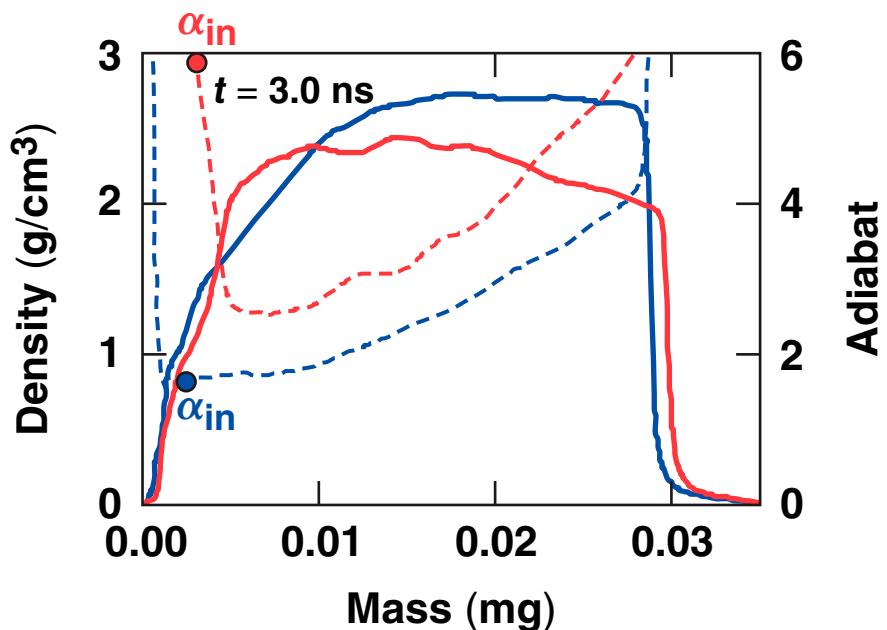
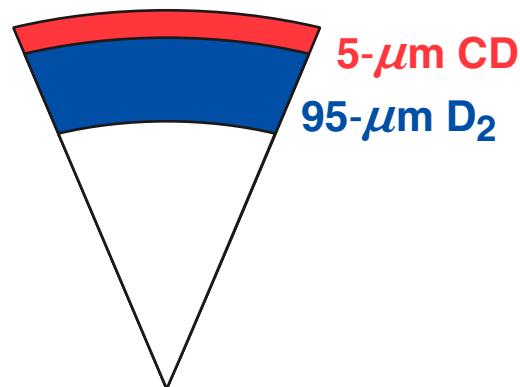


## Shock Timing

A correct thermal conduction model is essential  
for designing low-adiabat targets

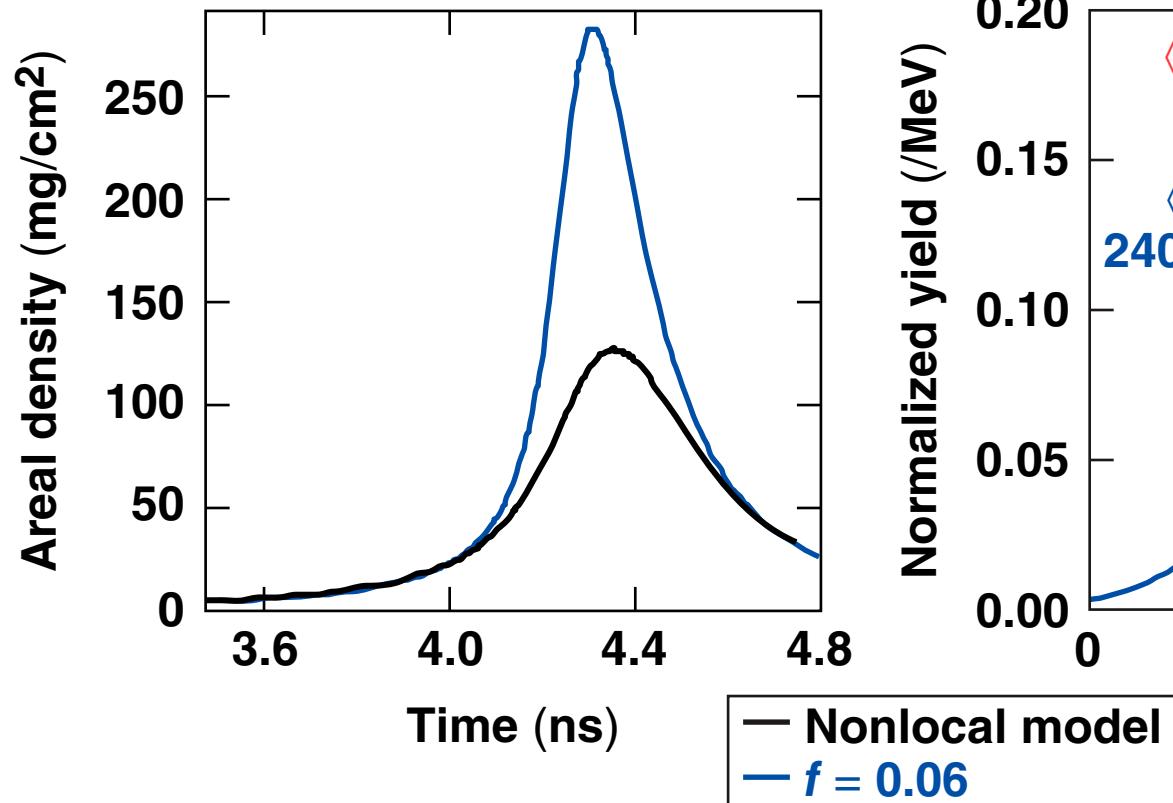


TC7811a



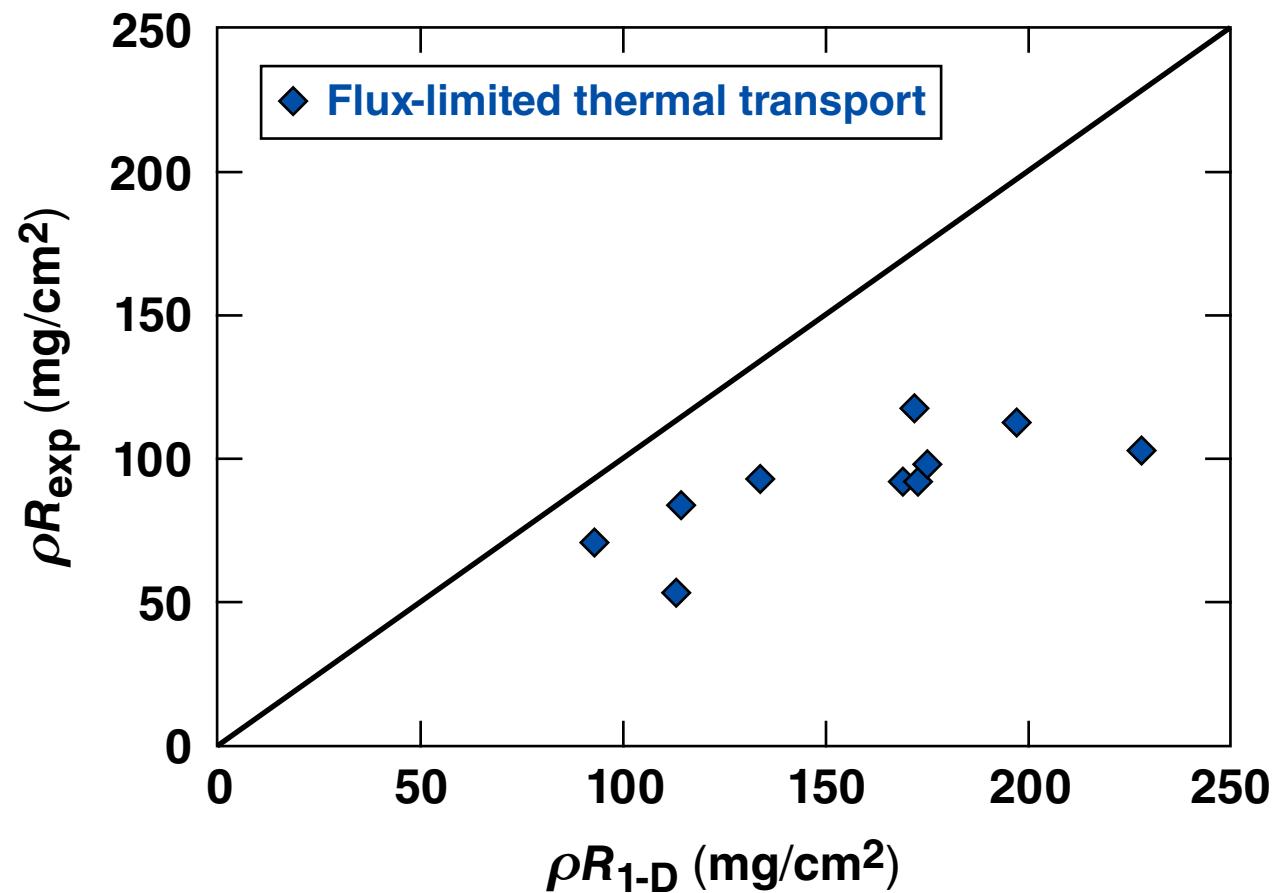
## Shock Timing

The areal density can be significantly changed depending on the thermal conduction model

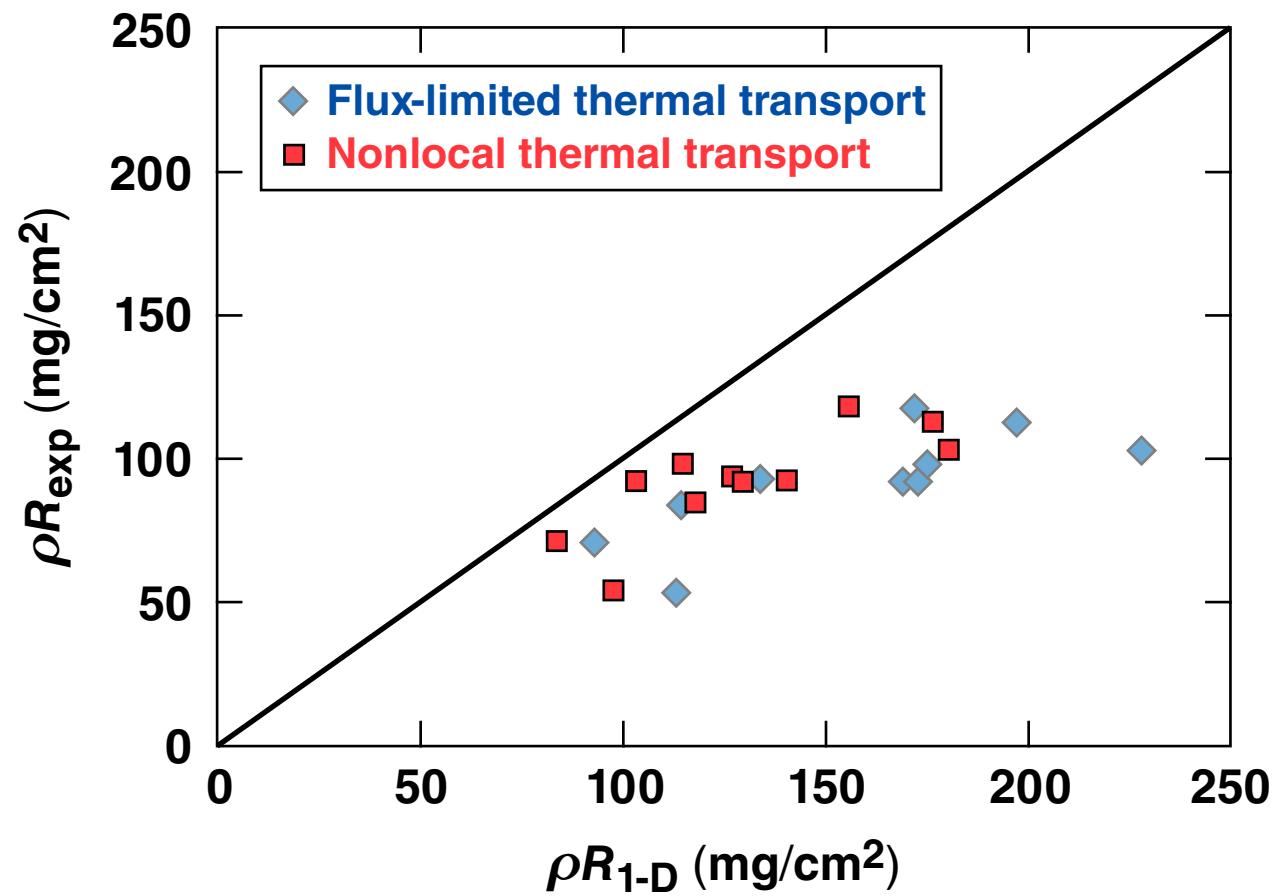


Shock timing must be adjusted using the nonlocal model.

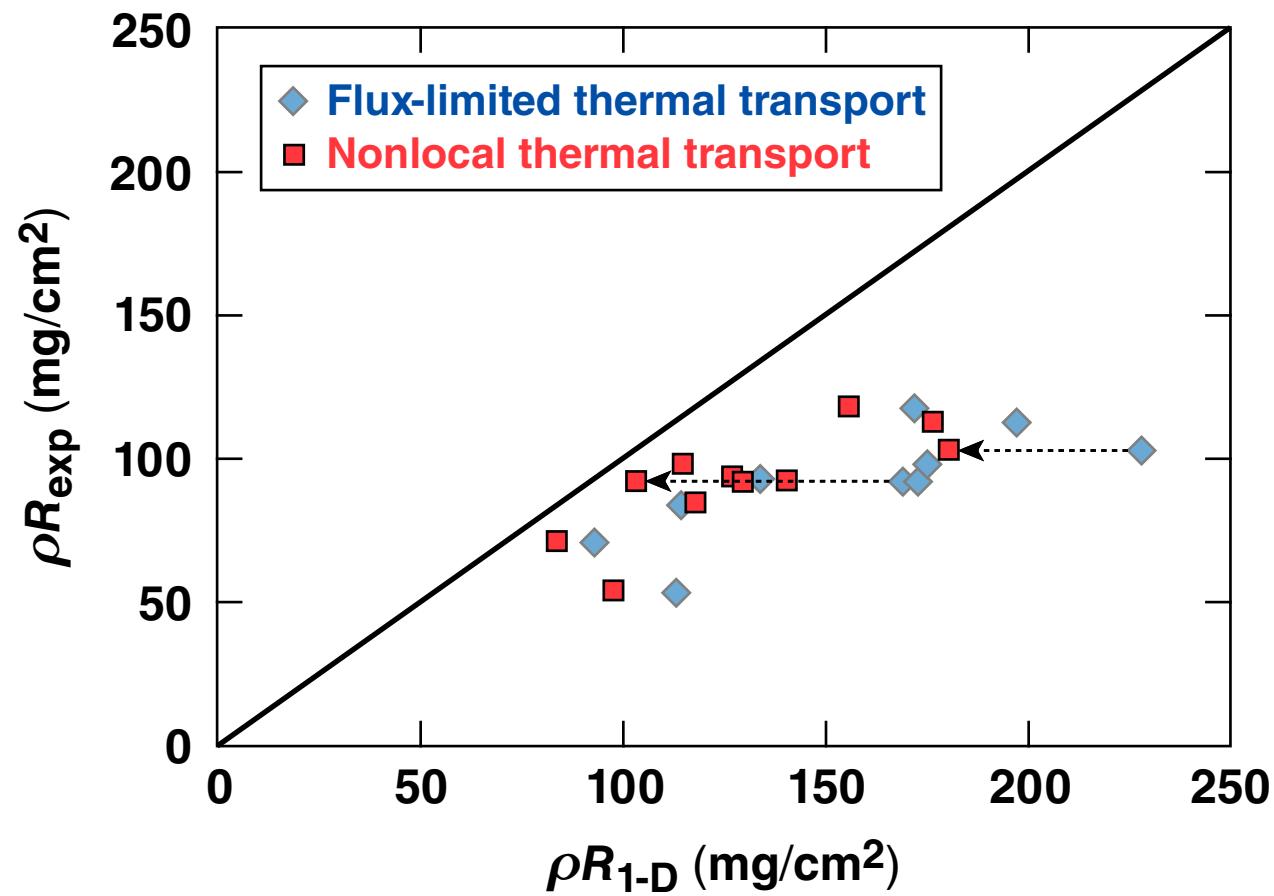
# The agreement between experimental data and code predictions improves when the nonlocal model is used



# The agreement between experimental data and code predictions improves when the nonlocal model is used

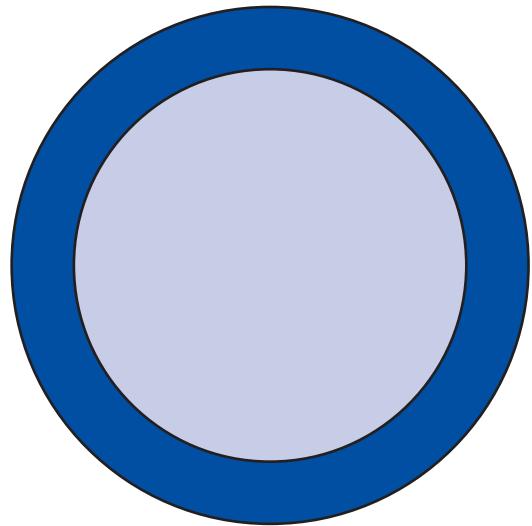


# The agreement between experimental data and code predictions improves when the nonlocal model is used



## Preheat

Hot-electron preheat generated by laser–plasma interaction can contribute to the final areal density degradation



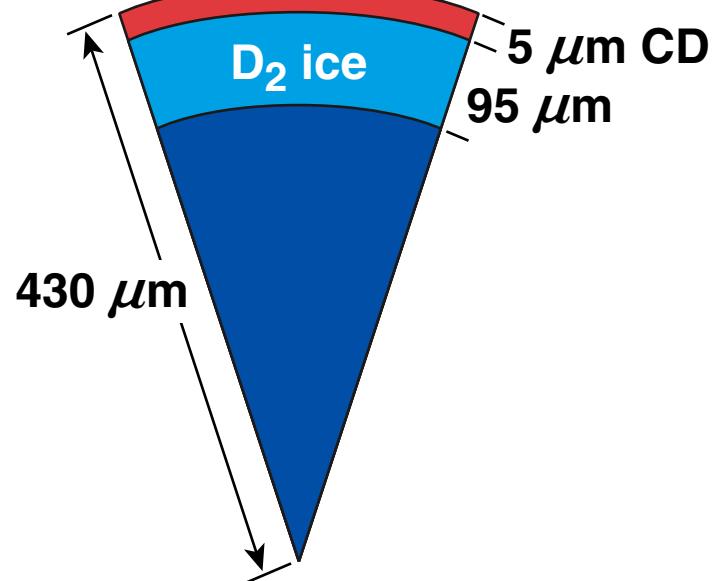
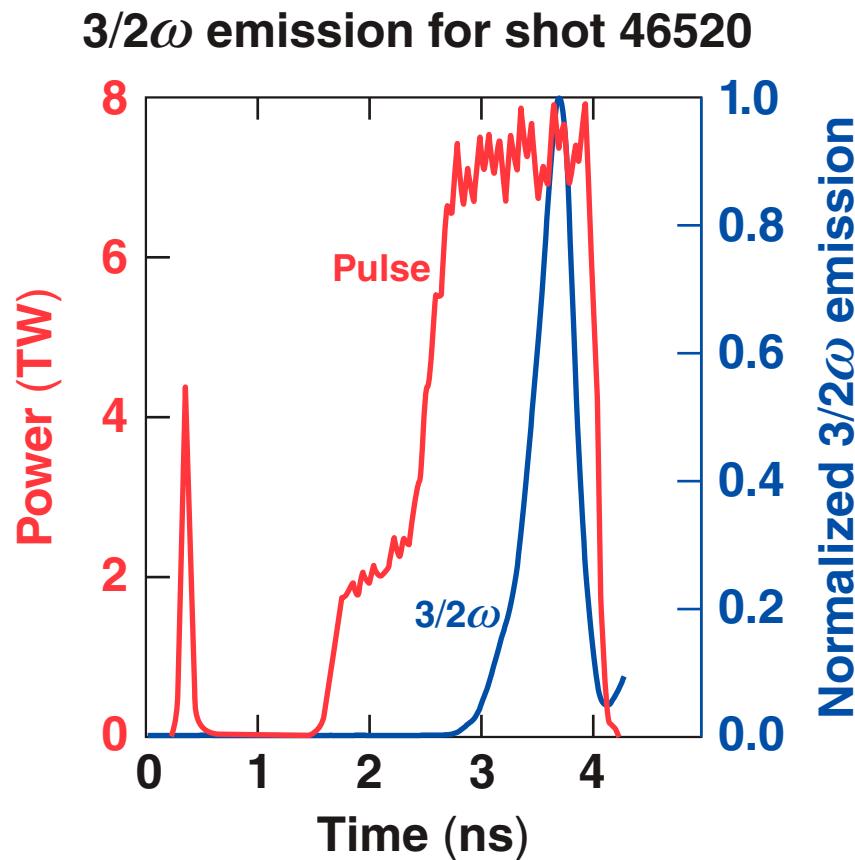
$$p \sim \alpha \rho^{5/3} \Rightarrow \alpha \sim \frac{T^{5/3}}{p^{2/3}}$$

$$\rho R \sim \alpha^{-0.54} \Rightarrow \rho R = \frac{\rho R_0}{\left(1 + \Delta T_{\text{preheat}} / T_0\right)^{0.9}}$$

- Low- $\alpha$  designs  $T_0 \sim 20$  eV
- 20%  $\rho R$  reduction for  $\Delta T_{\text{preheat}} \approx 6$  eV
- For OMEGA designs  $\frac{\rho R}{\rho R_0} \sim 0.8$  if  $E_{\text{preheat}} \sim 10$  J ( $< 0.1\% E_L$ )

## Preheat

$3/2\omega$  light indicates the presence  
of two-plasmon-decay instability

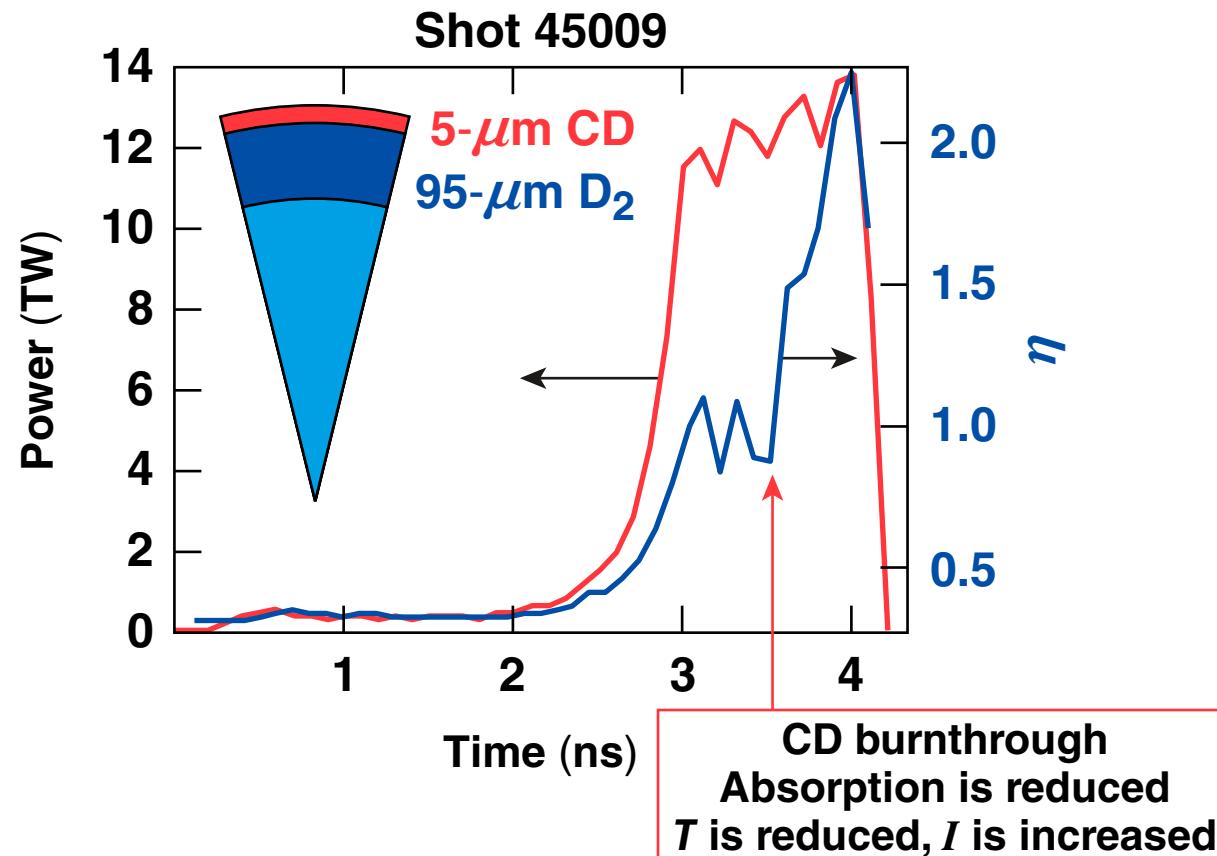


## Preheat

# Onset of $3/2\omega$ signal correlates with CD burnthrough time



- Above-threshold parameter\* for  $2\omega_p$  instability  $\eta = \frac{I_{14}L_{\mu m}}{230T_{keV}}$
- Instability develops when  $\eta > 1$



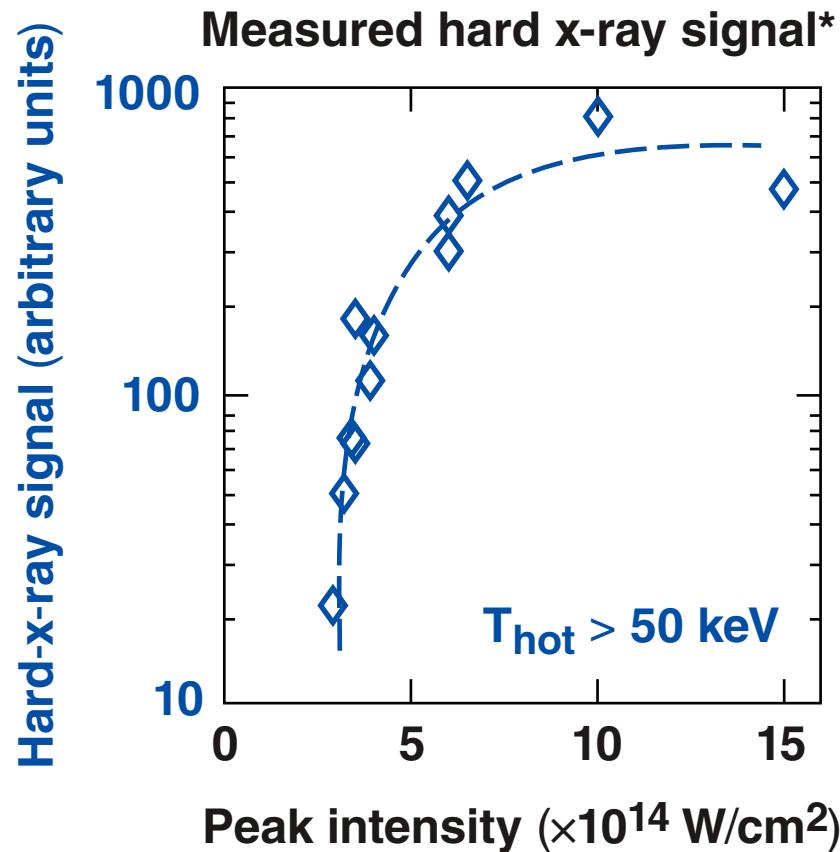
\*A. Simon et al., Phys. Fluids **26**, 3107 (1983).  
J. A. Delettrez (JO3.00003)

## Preheat

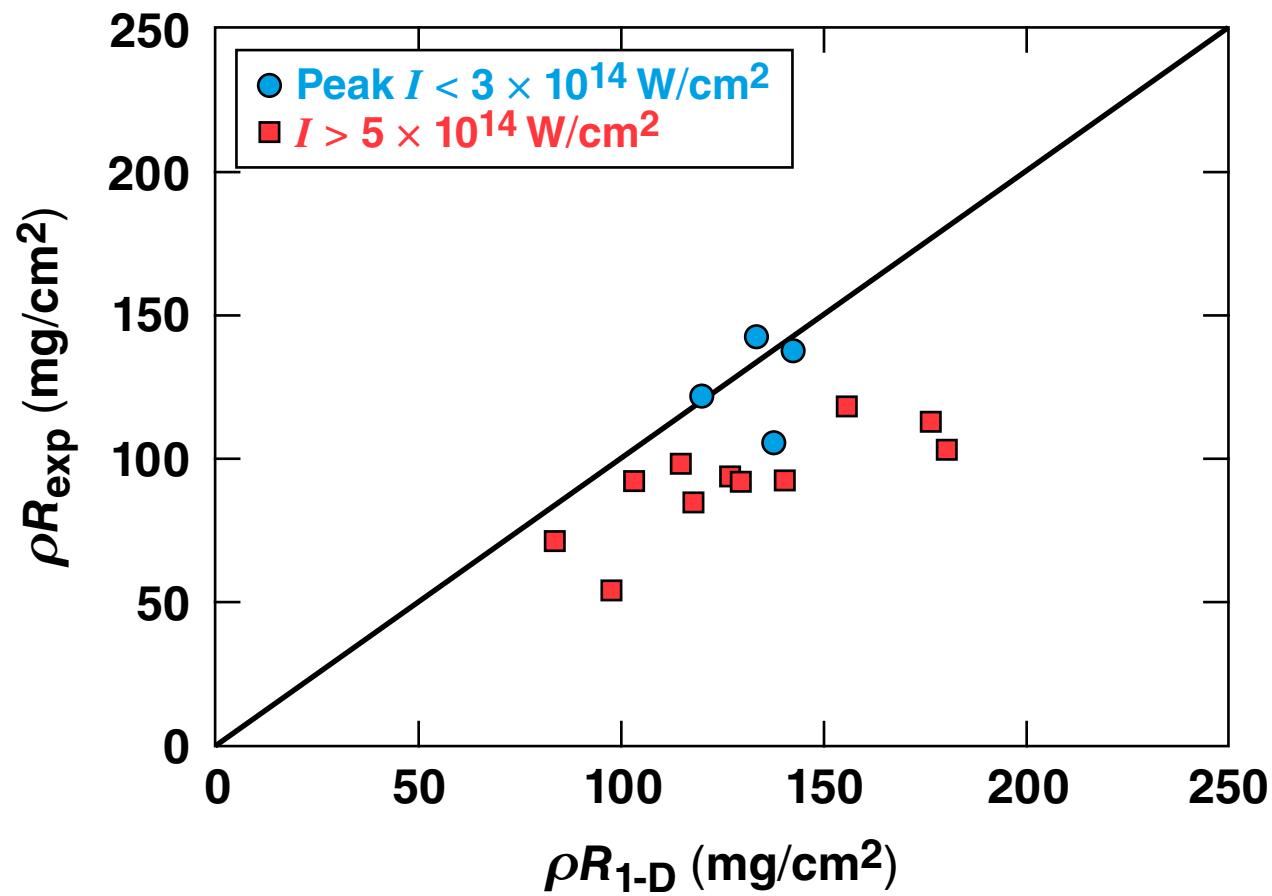
The measured hard x-ray signal  
is correlated with laser intensity



Cryogenic low- $\alpha$  D<sub>2</sub> targets with thin ( $\leq 5 \mu\text{m}$ ) CD ablator

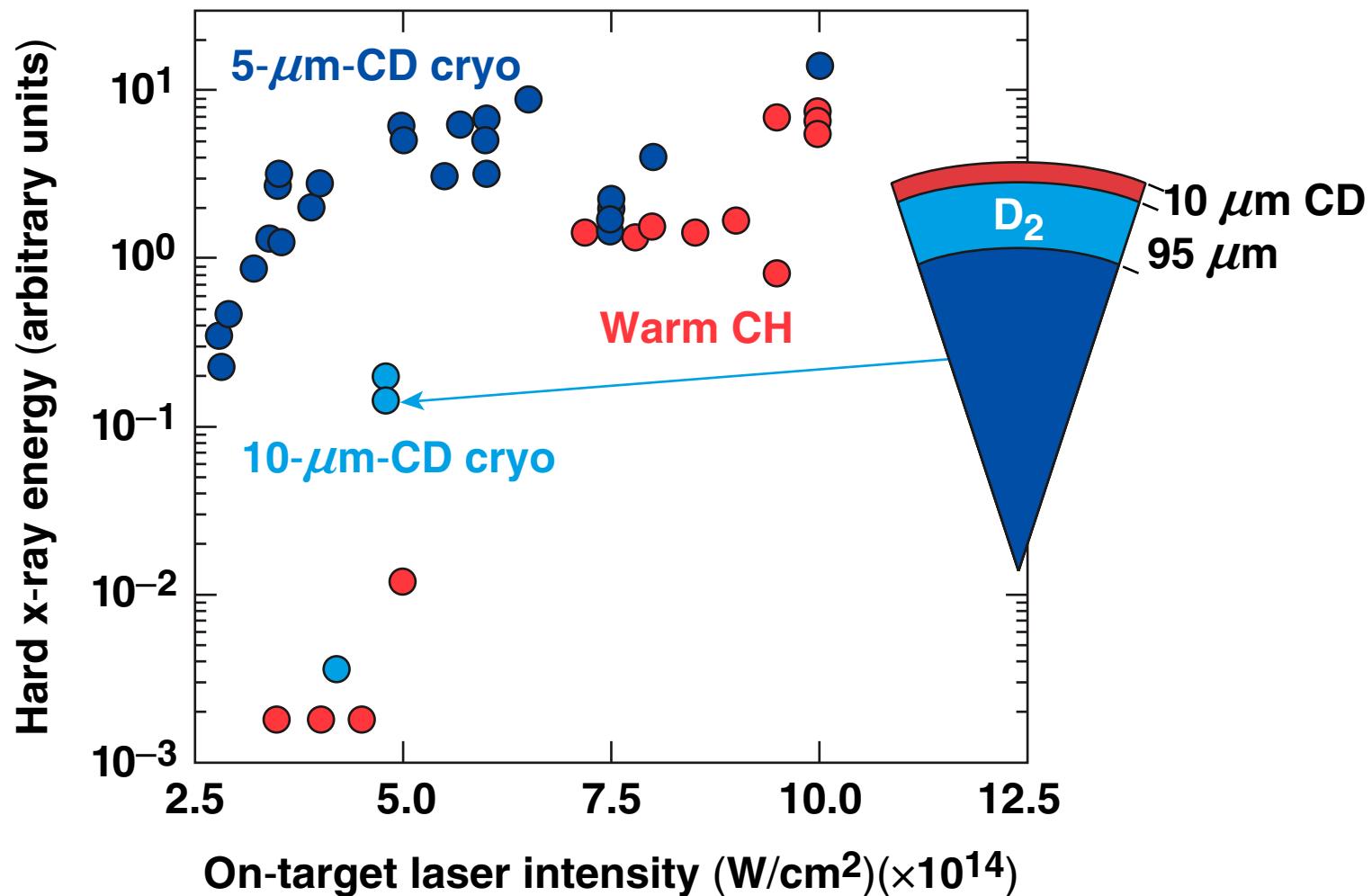


# An improved agreement between simulated and measured $\rho R$ is observed for low intensity implosions

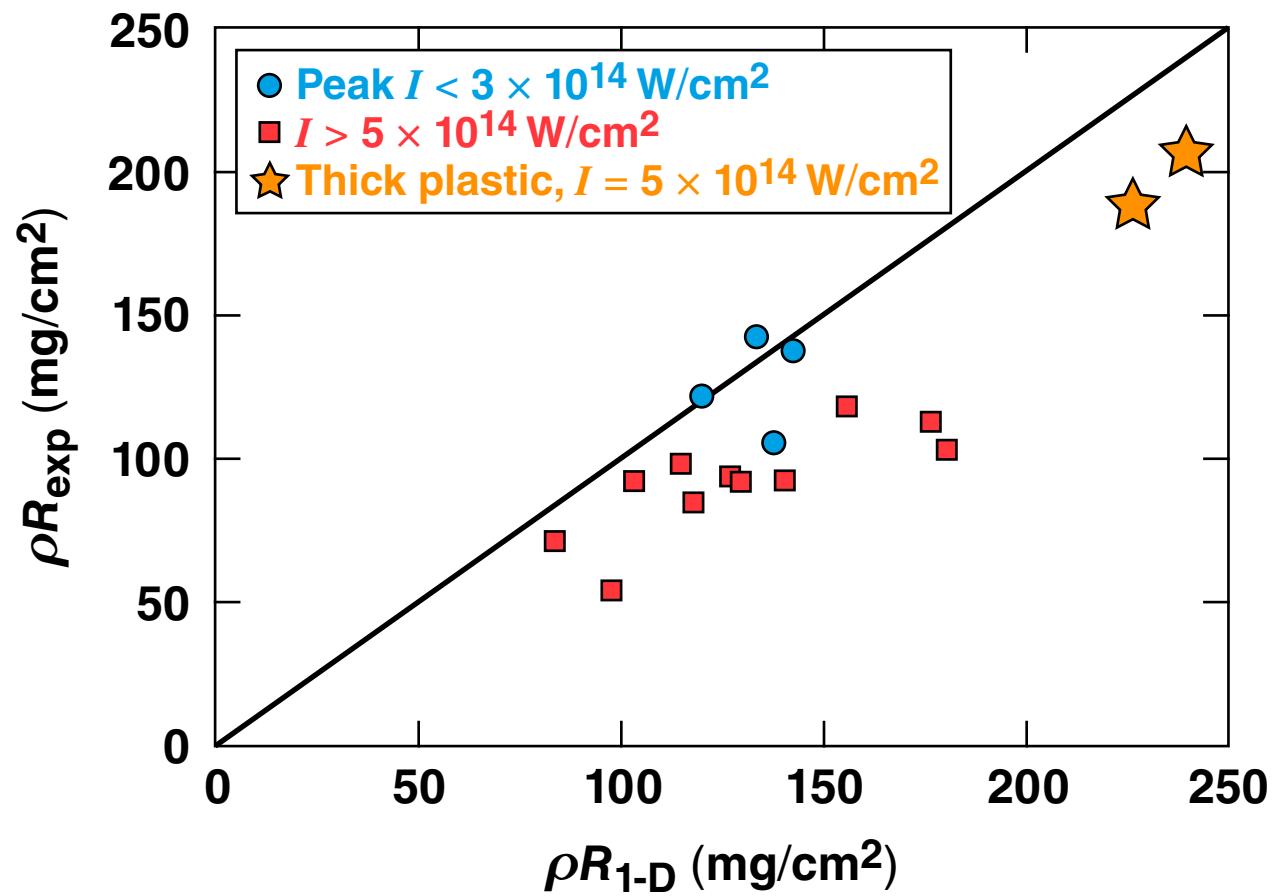


## Preheat

Suprathermal-electron preheat at higher intensities was mitigated by thickening the CD-overcoat layer



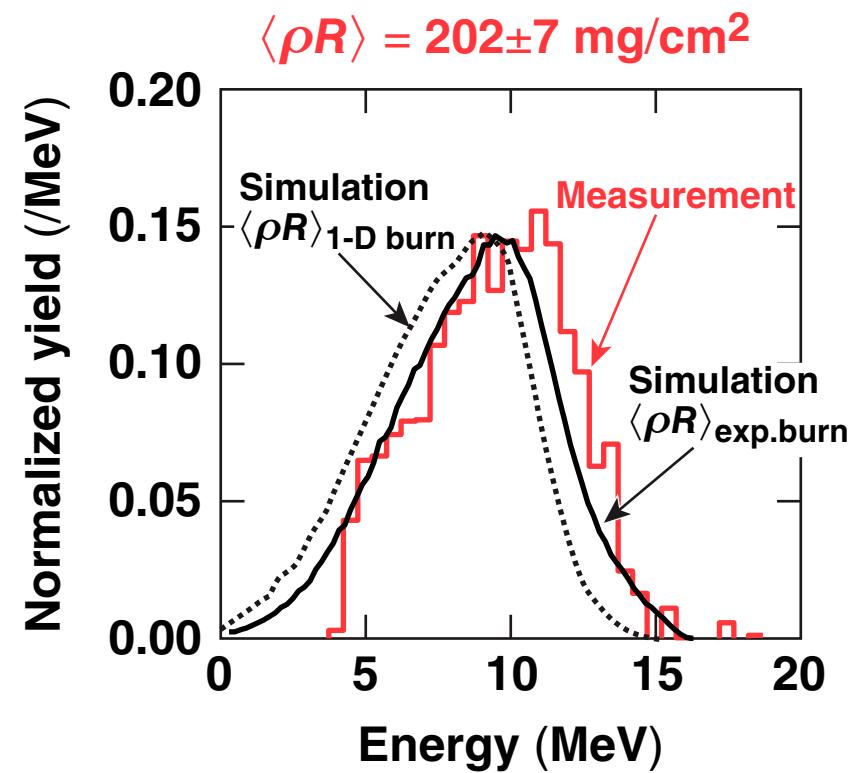
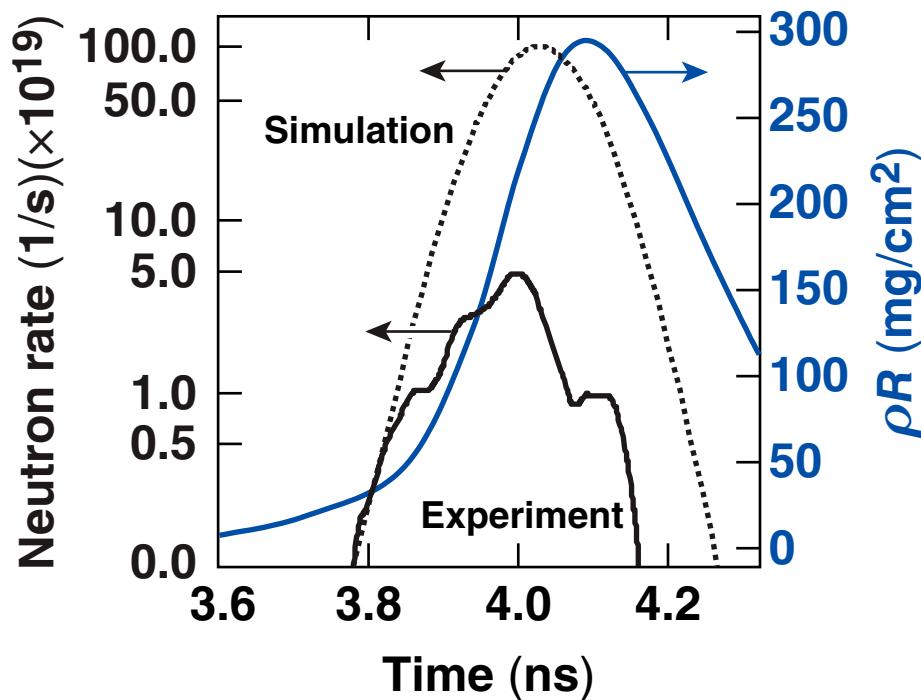
Areal density above 200 mg/cm<sup>2</sup> was achieved  
using 10-μm-thick CD ablators



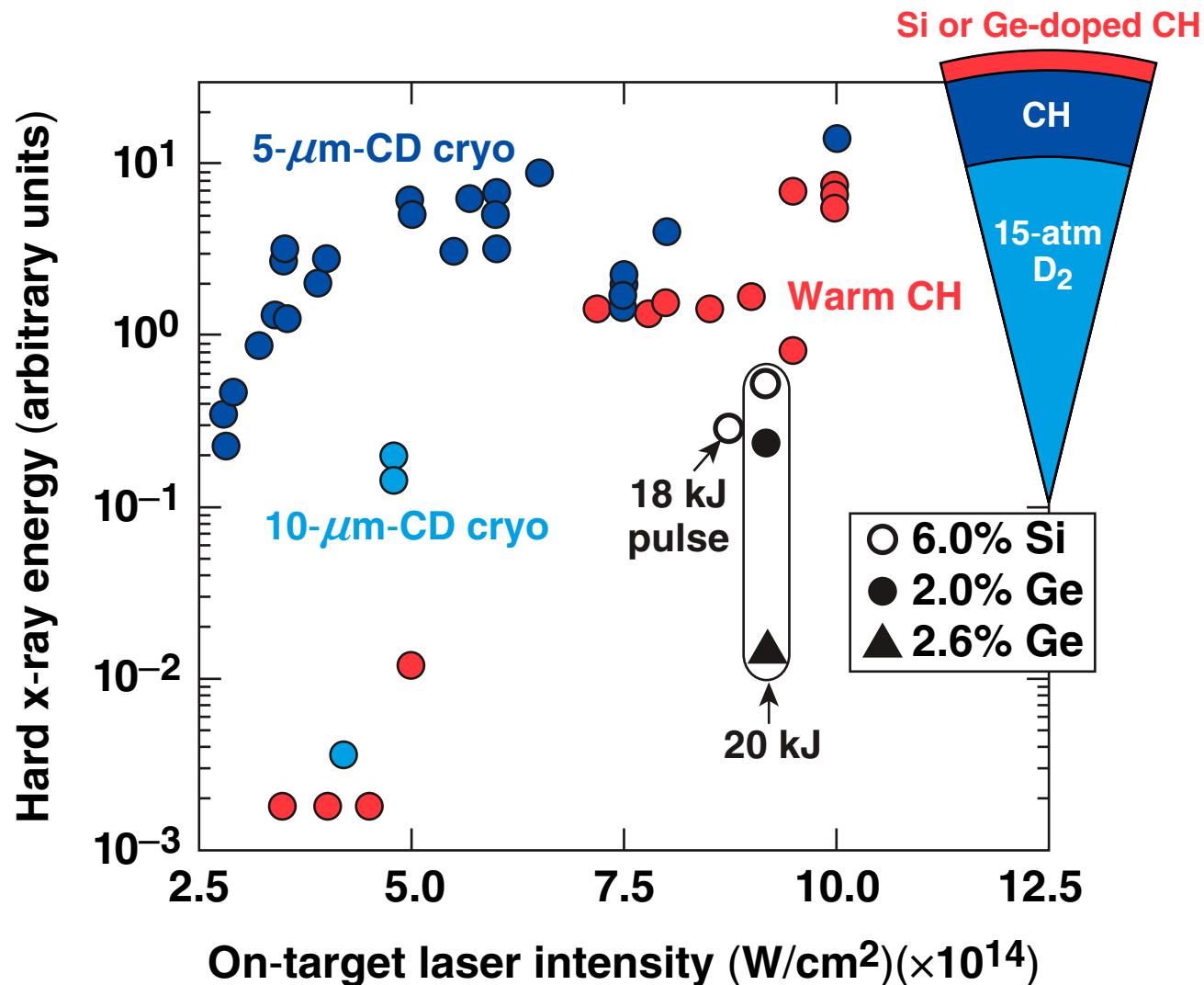
# Agreement with charge particle spectrum is improved when predicted $\rho R$ is averaged over the experimental burn history



- $\alpha = 2$  design



If required for future designs, the hot electron generation can be further reduced by using high-Z dopants

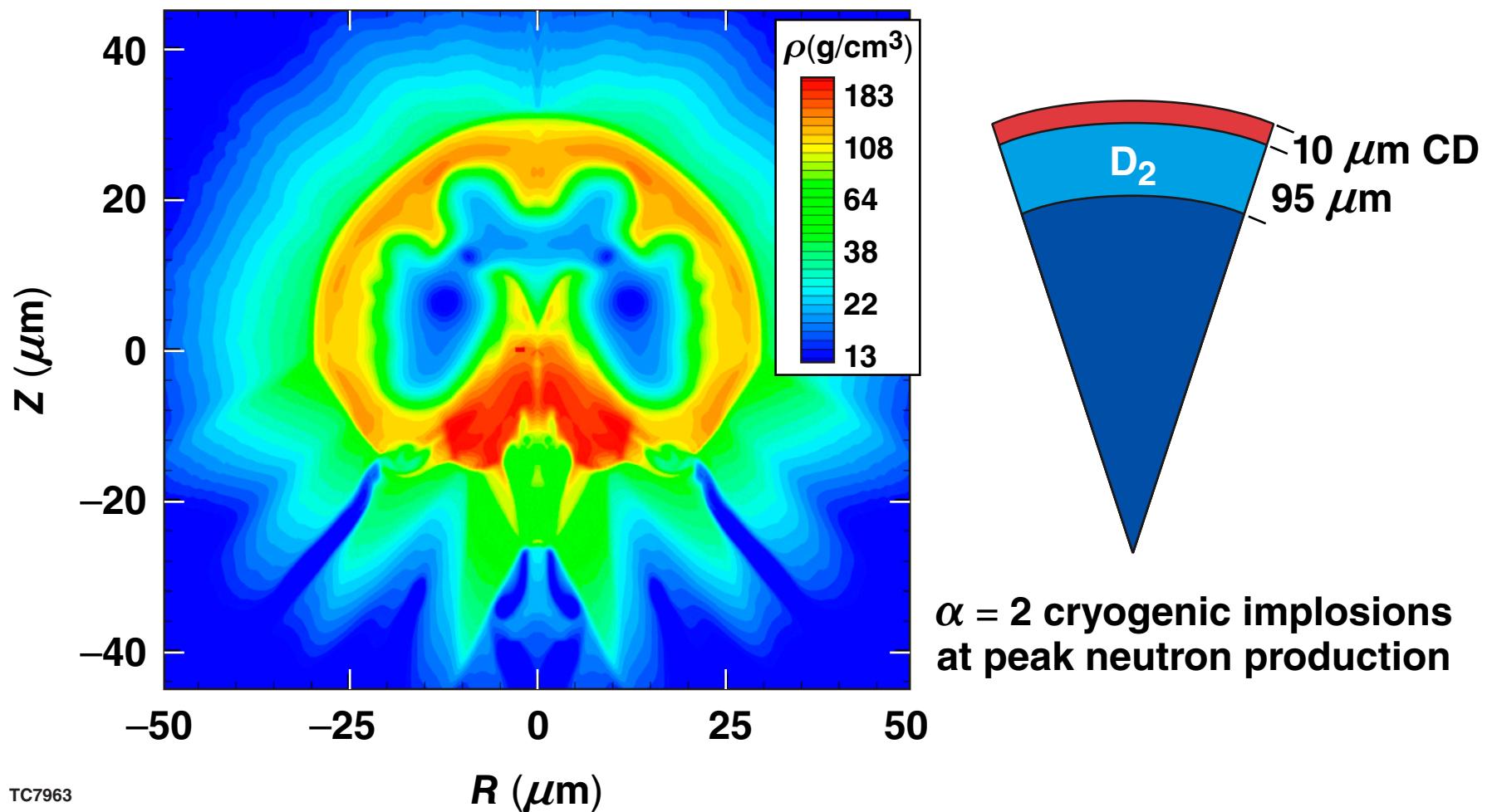


## Nonuniformity growth

**DRACO simulations predict significant shell deformation  
in low-adiabat, thick-CD cryogenic implosions**

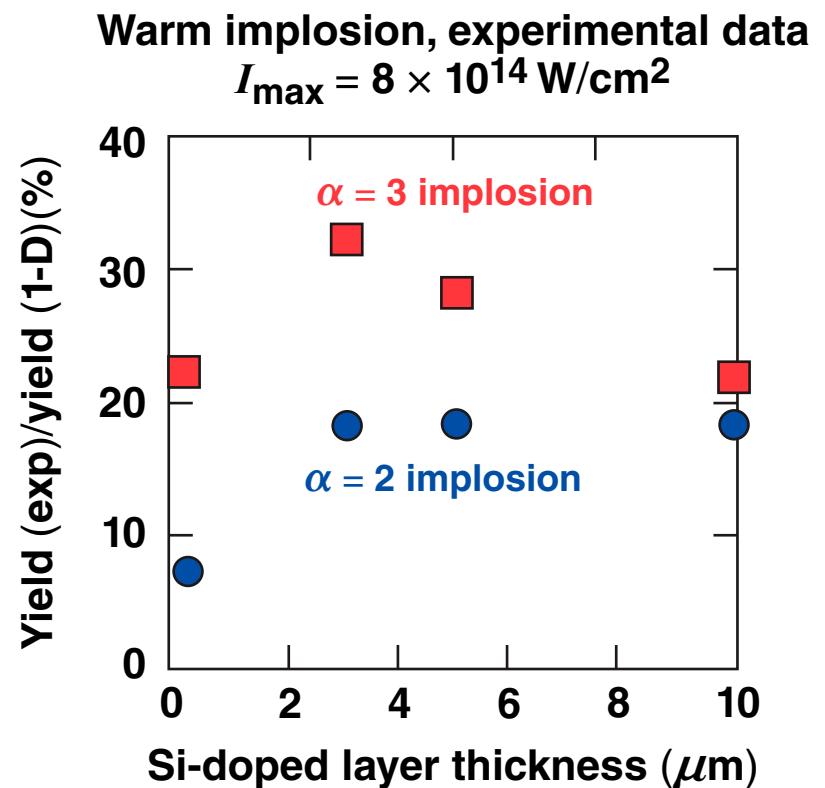
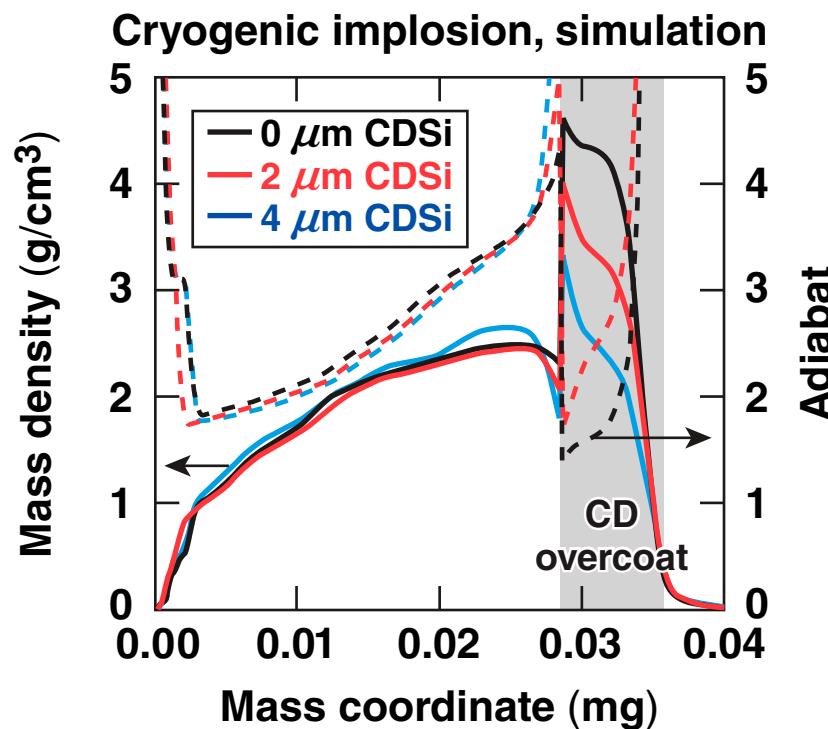


- Thick-CD designs have lower ablation velocity and increased RT growth factors



## Nonuniformity growth

# High-Z-doped ablators help reducing the laser imprint<sup>1,2</sup> and Rayleigh-Taylor instability growth<sup>3</sup>



<sup>1</sup>M. Karasik *et al.*, Bull. Am. Phys. Soc. **49**, 276 (2004).

<sup>2</sup>A. Mostovich *et al.*, “Enhanced Direct-drive implosion with thin high-Z ablation layers,” submitted to Phys. Rev. Lett.

<sup>3</sup>J. P. Knauer (PO6.00010)

## Summary/Conclusions

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